Perturbation theory, KAM theory and Celestial Mechanics 9. Breakdown and applications

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- 1. Break-down of quasi-periodic tori and attractors
- 2. KAM break-down criterion
- 3. Partial justification of Greene's method
- 4. Complex perturbing parameter
- 5. Applications

1. Break-down of quasi-periodic tori and attractors

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• We can compute a rigorous lower bound of the break–down threshold of invariant tori by means of KAM theory.

• Which is the real break-down value?

• In physical problems one can compare KAM result with a measure of the parameter. For example in the 3-body problem, $\varepsilon = \frac{m_{Jupiter}}{m_{Sum}} \simeq 10^{-3}$.

• In model problems one needs to apply numerical techniques: KAM break–down criterion, Greene's technique, frequency analysis, etc.

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KAM break-down criterion [Calleja, Celletti 2010]

• Solve the invariance equation for (K, μ) :

$$f_{\mu} \circ K(\theta) = K(\theta + \omega)$$
.

• Numerically efficient criterion: close to breakdown, one has a blow up of the Sobolev norms of a trigonometric approximation of the embedding:

$$K^{(L)}(\theta) = \sum_{|\ell| \le L} \widehat{K}_{\ell} e^{i\ell\theta} .$$

• A regular behavior of $||K^{(L)}||_m$ as ε increases (for λ fixed) provides evidence of the existence of the invariant attractor. Table: ε_{crit} for $\omega_r = 2\pi \frac{\sqrt{5}-1}{2}$.

Conservative case	Dissipative case	
ε_{crit}	λ	ε_{crit}
0.9716	0.9	0.9721
	0.5	0.9792

Greene's method, periodic orbits and Arnold's tongues

• Greene's method: breakdown of $\mathcal{C}(\omega)$ related to the stability of $\mathcal{P}(\frac{p_j}{q_j}) \to \mathcal{C}(\omega)$, but in the dissipative case: drift in an interval - *Arnold tongue* - admitting a periodic orbit.

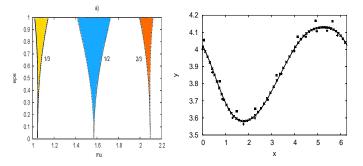


Figure: Left: Arnold's tongues providing μ vs. ε for 3 periodic orbits. Right: For $\lambda = 0.9$ and $\varepsilon = 0.5$ invariant attractor with frequency ω_r and approximating periodic orbits: 5/8 (*), 8/13 (+), 34/55 (×).

p_j/q_j	$\varepsilon_{p_j,q_j}^{\omega_r}(cons)$ $\varepsilon_{Sob} = [0.9716]$	$\varepsilon_{p_j,q_j}^{\omega_r}(\lambda=0.9)$ $\varepsilon_{Sob} = [0.972]$	$\varepsilon_{p_j,q_j}^{\omega_r}(\lambda=0.5)$
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$\begin{vmatrix} p_j/q_j \\ \varepsilon_{Sob} = \begin{bmatrix} 0.9 \end{bmatrix}$	$\begin{array}{c} \text{ns})\\ \text{0716} \end{array} \begin{vmatrix} \varepsilon_{p_j,q_j}^{\omega_r} (\lambda = 0.97) \\ \varepsilon_{Sob} = [0.97] \end{vmatrix}$	$\begin{array}{c} 9) \\ \varepsilon_{p_j,q_j}^{\omega_r}(\lambda = 0.5) \\ \varepsilon_{Sob} = [0.979] \end{array}$
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Partial justification of Greene's method [Calleja, Celletti, de la Llave, Falcolini 2014]

• Greene's criterion: originally developed for the standard map, gives the existence of an invariant curve with frequency ω if and only if the periodic orbits with frequencies given by the rational approximants p_j/q_j approaching ω are at the border of linear stability, measured by the *residue* $R(\frac{p_j}{q_i}) = \frac{1}{4}(2 - Tr(Df^q)).$

• Partial justifications for the symplectic case (Falcolini–de la Llave, MacKay) show that all periodic orbits with rotation number close to ω will have small residue.

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• We use the linearization theorem and give 2 different proofs: deformation theory and NHIM theory.

• Let the periodic orbit have frequency $\nu = (a_1, ..., a_n)/L$ with $a_j \in \mathbb{Z}, L \in \mathbb{N}$. The spectrum has a pairing rule: $Spec(Df^L) = \{\gamma_i, \lambda^L \gamma_i^{-1}\}.$

Theorem (Calleja, A.C., Falcolini, de la Llave, 2013)

Let f_{μ} be conformally symplectic, such that f_0 admits a Lagrangian invariant torus with frequency ω . Then, there exists a ngh. \mathcal{U} of the torus, s.t. when the periodic orbit with $\nu = (a_1, ..., a_n)/L$ is in \mathcal{U} , there exists $C_N > 0$ s.t.

$$|\gamma_i - 1| \le L C_N ||\mu||^N \simeq C_N ||\omega - \nu||^N$$
, $i = 1, ..., n$.

- Thus we have bounds on the spectral numbers of the periodic orbits.
- We get also upper/lower bounds on the width of the Arnold tongues.

- Proof: deformation theory:
- Find a smooth change of variables (normal form) that reduces the system to $(\theta + S_{\mu}, \lambda I)$ up to an error $(S_{\mu}$ polynomial function)
- ► The spectrum is invariant under smooth changes of variables
- ► For the system in normal form neglecting the remainder, the spectral numbers are equal to 1 and the residue is zero
- ► Estimate the spectrum by bounding the error in the normal form (use the theory of deformations, [de la Llave, Banyaga, Wayne, Marco, Moriyón]).

• Proof: NHIM and averaging theory:

▶ NHIM theory (Fenichel, Hirsch, Pugh, Shub): \mathcal{T}_{μ} is a family of tori invariant under f_{μ} (the invariant torus for f_0 is a NHIM)

► We can write these manifolds as the image of the torus under a family of maps K_{μ} such that $f_{\mu} \circ K_{\mu} = K_{\mu} \circ R_{\mu}$, where R_{μ} denotes the dynamics of f_{μ} restricted to \mathcal{T}_{μ} (R_0 is the Diophantine rotation)

Averaging theory tells us that for $N \le N_0$ we can find a diffeomorphism $B_{\mu}^{(N)}$ and a rotation $T_{\omega_{\mu}^{(N)}}$ such that

$$(B^{(N)}_{\mu})^{-1} \circ R_{\mu} \circ B^{(N)}_{\mu} = T_{\omega^{(N)}_{\mu}} + O(\|\mu\|^{N+1})$$

▶ Periodic orbits (in the NHIM) have *n* Lyapunov exponents close to 1

▶ Pairing rule and Lagrangian character of the tori imply that the remaining exponents of the periodic orbit with $\rho = a/L$ are close to λ^L .

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• We compute the solution of the functional equation assuming $\varepsilon \in \mathbb{C}$. Applying Newton's method we follow the solution from $\varepsilon = 0$ increasing the real and imaginary parts of $\varepsilon = \varepsilon_r + i\varepsilon_i$ until blow-up.

• The expansion of the parametrization *K* in terms of the complex ε as the sum of a real and an imaginary part becomes $(K_j(\theta) \text{ are real})$

$$\begin{aligned} K(\theta;\varepsilon) &= \sum_{j=1}^{\infty} K_j(\theta) (\varepsilon_r + i\varepsilon_i)^j \\ &= K_r(\theta;\varepsilon_r,\varepsilon_i) + iK_i(\theta;\varepsilon_r,\varepsilon_i) \end{aligned}$$

and the same for $g(\theta + K) = \sin(\theta + K)$:

$$\varepsilon g(\theta + K) = \varepsilon_r g_r - \varepsilon_i g_i + i(\varepsilon_r g_i + \varepsilon_i g_r).$$

• Setting $\gamma = \omega(1 - \lambda) - \mu = \gamma_r + i\gamma_i$, the functional equation corresponds to the following two equations:

$$D_1 D_\lambda K_r(\theta + \omega; \varepsilon_r, \varepsilon_i) - \varepsilon_r g_r(\theta) + \varepsilon_i g_i(\theta) - \gamma_r = 0$$

$$D_1 D_\lambda K_i(\theta + \omega; \varepsilon_r, \varepsilon_i) - \varepsilon_r g_i(\theta) - \varepsilon_i g_r(\theta) - \gamma_i = 0.$$

• Figure: domains of existence in the complex ε -plane for different mappings, for $\omega/(2\pi) = [3, 12, 1, 1, 1, 1, ...]$ and the golden ratio, for specific values of λ (cut of Figure top-right is possibly due to the fact that the frequency is close to a rational).

• The shapes of the existence domains strongly depend on the choice of the function $g(\theta)$ (bottom panel).

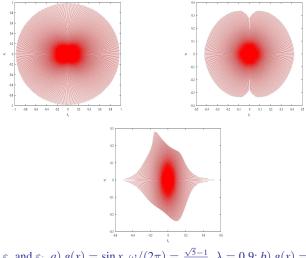


Figure: Axes: ε_r and ε_i . a) $g(x) = \sin x$, $\omega/(2\pi) = \frac{\sqrt{5}-1}{2}$, $\lambda = 0.9$; b) $g(x) = \sin x$, $\omega/(2\pi) = [3, 12, 1, 1, 1, 1, ...]$, $\lambda = 0.9$; c) $g(x) = \sin x + \frac{1}{20} \sin(4x) + \frac{1}{30} \sin(6x)$, $\omega/(2\pi) = \frac{\sqrt{5}-1}{2}$, $\lambda = 0.9$.

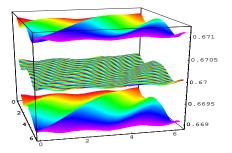
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Applications

- Standard map
- Rotational dynamics: (spin-orbit problem)
- Orbital dynamics: (three–body problem)

KAM stability through confinement

• Confinement in 2-dimensional systems: dim(phase space)=4, dim(constant energy level)=3, dim(invariant tori)=2 \rightarrow confinement in phase space for ∞ times between bounding invariant tori



• Confinement no more valid for n > 2: the motion can diffuse through invariant tori, reaching arbitrarily far regions (Arnold's diffusion).

Conservative standard map

Results of the '90s

• [A.C., L. Chierchia] Let $\omega = 2\pi \frac{\sqrt{5}-1}{2}$; $|\varepsilon| \le 0.838$ (86% of Greene's value) there exists an invariant curve with frequency ω .

• [R. de la Llave, D. Rana] Using accurate strategies and efficient computer–assisted algorithms, the result was improved to 93% of Greene's value.

• Very recent results [J.-L. Figueras, A. Haro, A. Luque] in

http://arxiv.org/abs/1601.00084 reaching 99.9%!!!

Dissipative standard map

• Using $K_2(\theta) = \theta + u(\theta)$, the invariance equation is

$$D_1 D_\lambda u(\theta) - \varepsilon \sin(\theta + u(\theta)) + \omega(1 - \lambda) - \mu = 0$$
(1)

with $D_{\lambda}u(\theta) = u(\theta + \frac{\omega}{2}) - \lambda u(\theta - \frac{\omega}{2}).$

Proposition [dissipative standard map, R. Calleja, A.C., R. de la Llave (2016)]

Let $\omega = 2\pi \frac{\sqrt{5}-1}{2}$ and $\lambda = 0.9$; then, for $\varepsilon \leq \varepsilon_{KAM}$, there exists a unique solution $u = u(\theta)$ of (1), provided that $\mu = \omega(1 - \lambda) + \langle u_{\theta} D_1 D_{\lambda} u \rangle$.

• The drift μ must be suitably tuned and cannot be chosen independently from ω .

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• The drift μ must be suitably tuned and cannot be chosen independently from ω .

• Preliminary result: conf. symplectic version, careful estimates, continuation method using the Fourier expansion of the initial approximate solution \Rightarrow

 $\varepsilon_{KAM} = (99\% \text{ of the critical breakdown threshold })$

Rotational dynamics

The **Moon** and all evolved satellites, always point the same face to the host planet: 1:1 resonance, i.e. 1 rotation = 1 revolution (Phobos, Deimos - Mars, Io, Europa, Ganimede, Callisto - Jupiter, Titan, Rhea, Enceladus, etc.). Only exception: **Mercury** in a 3:2 spin–orbit resonance (3 rotations = 2 revolutions).

• Important dissipative effect: **tidal torque**, due to the non-rigidity of planets and satellites.

Conservative spin-orbit problem

• Spin–orbit problem: triaxial satellite S (with A < B < C) moving on a Keplerian orbit around a central planet \mathcal{P} , assuming that the spin–axis is perpendicular to the orbit plane and coincides with the shortest physical axis.

Conservative spin-orbit problem

Spin–orbit problem: triaxial satellite S (with A < B < C) moving on a Keplerian orbit around a central planet P, assuming that the spin–axis is perpendicular to the orbit plane and coincides with the shortest physical axis.
Equation of motion:

$$\ddot{x} + \varepsilon (\frac{a}{r})^3 \sin(2x - 2f) = 0$$
, $\varepsilon = \frac{3}{2} \frac{B - A}{C}$.

• The (Diophantine) frequencies of the bounding tori are for example:

$$\omega_{-} \equiv 1 - \frac{1}{2 + \frac{\sqrt{5}-1}{2}} , \qquad \omega_{+} \equiv 1 + \frac{1}{2 + \frac{\sqrt{5}-1}{2}} .$$

Proposition [spin-orbit model, A.C. (1990)]

Consider the spin–orbit Hamiltonian defined in $U \times \mathbb{T}^2$ with $U \subset \mathbb{R}$ open set. Then, for the true eccentricity of the Moon e = 0.0549, there exist invariant tori, bounding the motion of the Moon, for any $\varepsilon \leq \varepsilon_{Moon} = 3.45 \cdot 10^{-4}$. • Possible forthcoming estimates: spin–orbit equation with tidal torque given by

$$\ddot{x} + \varepsilon \left(\frac{a}{r}\right)^3 \sin(2x - 2f) = -\lambda(\dot{x} - \mu) , \qquad (2)$$

where λ , μ depend on the orbital (e) and physical properties of the satellite.

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Proposition [A.C., L. Chierchia (2009)]

Let $\lambda_0 \in \mathbb{R}_+$, ω Diophantine. There exists $0 < \varepsilon_0 < 1$, such that for any $\varepsilon \in [0, \varepsilon_0]$ and any $\lambda \in [-\lambda_0, \lambda_0]$ there exists a unique function $u = u(\theta, t)$ with $\langle u \rangle = 0$, such that

$$x(t) = \omega t + u(\omega t, t)$$

solves the equation of motion with $\mu = \omega (1 + \langle u_{\theta}^2 \rangle)$.

Conservative three–body problem

- Consider the motion of a small body (with negligible mass) under the gravitational influence of two primaries, moving on Keplerian orbits about their common barycenter (*restricted* problem).
- Assume that the orbits of the primaries are circular and that all bodies move on the same plane: *planar, circular, restricted three–body problem* (PCR3BP).

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Adopting suitable normalized units and action–angle Delaunay variables (L, G) ∈ ℝ², (ℓ, g) ∈ T², we obtain a 2 d.o.f. Hamiltonian function:

$$\mathcal{H}(L,G,\ell,g) = -\frac{1}{2L^2} - G + \varepsilon R(L,G,\ell,g) \; .$$

• ε primaries' mass ratio ($\varepsilon = 0$ Keplerian motion). Actions: $L = \sqrt{a}$, $G = L\sqrt{1 - e^2}$.

• Degenerate Hamiltonian, but Arnold's isoenergetic non-degenerate (persistence of invariant tori on a fixed energy surface), i.e. setting $h(L, G) = -\frac{1}{2L^2} - G$:

$$\det \begin{pmatrix} h''(L,G) & h'(L,G) \\ h'(L,G)^T & 0 \end{pmatrix} = \det \begin{pmatrix} -\frac{3}{L^4} & 0 & \frac{1}{L^3} \\ 0 & 0 & -1 \\ \frac{1}{L^3} & -1 & 0 \end{pmatrix} = \frac{3}{L^4} \neq 0 \quad \text{for all } L \neq 0.$$

• Dimension phase space = 4, fix the energy: dim = 3; dimension invariant tori = 2.

Result: The stability of the small body can be obtained by proving the existence of invariant surfaces which confine the motion of the asteroid on a preassigned energy level.

Sample: Sun, Jupiter, asteroid 12 Victoria with

$$a_{\rm V} \simeq 0.449$$
, $e_{\rm V} \simeq 0.220$, $v_{\rm V} \simeq \frac{8.363 - 1.305}{360} = 1.961 \cdot 10^{-2}$.

• Size of the perturbing parameter: $\varepsilon_J = 0.954 \cdot 10^{-3}$.

• Approximations: disregard $e_J = 4.82 \cdot 10^{-2}$ (worst physical approximation), inclinations, gravitational effects of other bodies (Mars and Saturn), dissipative phenomena (tides, solar winds, Yarkovsky effect,...)

Empirical criterion: expand the perturbation in e and a, neglecting contributions smaller than e_J. Neglect terms of order O(ε) in F_ε (i.e. replace F_ε by F₀).
One-parameter family of Hamiltonians (0 < G < L):

$$H_{\mathrm{SJV}}(L,G,\ell,g;\varepsilon) = -rac{1}{2L^2} - G + \varepsilon H_1(L,G,\ell,g) \; ,$$

with $(a = L^2, e = \sqrt{1 - \frac{G^2}{L^2}})$

$$\begin{split} H_1(L,G,\ell,g) &= -(1+\frac{a^2}{4}+\frac{9}{64}a^4+\frac{3}{8}a^2e^2) \\ + & \left(\frac{1}{2}+\frac{9}{16}a^2\right)a^2e\,\cos\ell - \left(\frac{3}{8}a^3+\frac{15}{64}a^5\right)\cos(\ell+g) \\ + & \left(\frac{9}{4}+\frac{5}{4}a^2\right)a^2e\,\cos(\ell+2g) - \left(\frac{3}{4}a^2+\frac{5}{16}a^4\right)\cos(2\,\ell+2\,g) \\ - & \frac{3}{4}a^2e\,\cos(3\,\ell+2\,g) - \left(\frac{5}{8}a^3+\frac{35}{128}a^5\right)\cos(3\,\ell+3\,g) \\ - & \frac{35}{64}a^4\cos(4\,\ell+4\,g) - \frac{63}{128}a^5\cos(5\ell+5g) \,. \end{split}$$

• Fixing the perturbation parameter at the value $\varepsilon = \varepsilon_J$, we obtain the *Sun–Jupiter–Victoria Hamiltonian*:

$$\begin{aligned} \overline{H}_{\rm SJV}(L,G,\ell,g) &= -\frac{1}{2L^2} - G + \varepsilon_J H_1(L,G,\ell,g) , \\ &= H_0(L,G) + \varepsilon_J H_1(L,G,\ell,g) . \end{aligned}$$

• Observed values: $L_{\rm V} = \sqrt{a_{\rm V}} \simeq 0.670, G_{\rm V} = L_{\rm V} \sqrt{1 - e_{\rm V}^2} \simeq 0.654.$

• Define the "osculating energy value" in terms of the Keplerian approximation and in terms of the "secular" effects; define $E_V^{(0)}$ and $E_V^{(1)}$ as

$$\begin{aligned} H_0(L_V, G_V) &= -\frac{1}{2L_V^2} - G_V \simeq -1.768 = E_V^{(0)} , \\ \langle H_1(L_V, G_V, \cdot, \cdot) \rangle &\simeq -1.060 = E_V^{(1)} , \\ E_V(\varepsilon) &= E_V^{(0)} + \varepsilon E_V^{(1)} . \end{aligned}$$

• Osculating energy level of the Sun-Jupiter-Victoria model:

$$\overline{E}_{\mathrm{V}} = E_{\mathrm{V}}(\varepsilon_J) = E_{\mathrm{V}}^{(0)} + \varepsilon_J E_{\mathrm{V}}^{(1)} \simeq -1.769$$
.

• From now on we will be concerned with such one-parameter family of energy surfaces:

$$\mathcal{S}_{\varepsilon,\mathrm{V}} = H^{-1}_{\mathrm{SJV}}(E_{\mathrm{V}}(\varepsilon))$$

• We consider two invariant tori on $S_{0,V}$, which bound from above and below the observed value L_V : we define

$$ilde{L}_{\pm}=L_V\pm 0.001$$
 .

• The corresponding frequencies are:

$$\underline{\tilde{\omega}}_{\pm} = \frac{\partial H_0}{\partial (L,G)} = \left(\frac{1}{\tilde{L}_{\pm}^3}, -1\right) = (\tilde{\alpha}_{\pm}, -1) .$$

• Since we need Diophantine frequencies, we compute the continued fraction representation up to the order 5 of $\tilde{\alpha}_{\pm}$ and then we modify the frequencies by adding a tail of all one's.

• Result: two quadratic "noble" numbers α_{\pm} given by:

$$\begin{aligned} \alpha_{-} &= & [3; 3, 4, 2, 1^{\infty}] = 3.30976937631389...\\ \alpha_{+} &= & [3; 2, 1, 17, 5, 1^{\infty}] = 3.33955990647860.. \end{aligned}$$

We can now define the Diophantine frequencies

$$\underline{\omega}_{\pm} = (\alpha_{\pm}, -1) \; ,$$

with corresponding Diophantine constants

$$au_{\pm} = au = 1 \;, \quad \gamma_{-} = 7.224496 \cdot 10^{-3} \;, \quad \gamma_{+} = 3.324329 \cdot 10^{-2} \;.$$

• We are interested in the KAM continuation of the following unperturbed tori, which lie on the energy level $H_0^{-1}(E_V^{(0)})$:

$$\mathcal{T}_0^{\pm} = \{(L_{\pm}, G_{\pm})\} \times \mathbb{T}^2 ,$$

with

$$L_{\pm} = \frac{1}{\alpha_{\pm}^{1/3}}$$
, $G_{\pm} = -\frac{1}{2L_{\pm}^2} - E_V^{(0)}$.

Concrete example: Sun, Jupiter, asteroid 12 Victoria with a = 0.449 (in Jupiter–Sun unit distance) and e = 0.22, so that L_V ≃ 0.670, G_V ≃ 0.654.
Select the energy level as E^{*}_V = -¹/_{2L²} − G_V + ε_J⟨R(L_V, G_V, ℓ, g)⟩ ≃ −1.769,

where $\varepsilon_J \simeq 10^{-3}$ is the observed Jupiter–Sun mass–ratio. On such (3–dim) energy level prove the existence of two (2–dim) trapping tori with frequencies ω_{\pm} .

• Concrete example: Sun, Jupiter, asteroid 12 Victoria with a = 0.449 (in Jupiter–Sun unit distance) and e = 0.22, so that $L_V \simeq 0.670$, $G_V \simeq 0.654$. • Select the energy level as $E_V^* = -\frac{1}{2L_V^2} - G_V + \varepsilon_J \langle R(L_V, G_V, \ell, g) \rangle \simeq -1.769$, where $\varepsilon_J \simeq 10^{-3}$ is the observed Jupiter–Sun mass–ratio. On such (3–dim) energy level prove the existence of two (2–dim) trapping tori with frequencies ω_{\pm} .

Proposition [three-body problem, A.C., L. Chierchia (2007)]

Let $E = E_V^*$. Then, for $|\varepsilon| \le 10^{-3}$ the unperturbed tori with trapping frequencies ω_{\pm} can be analytically continued into KAM tori for the perturbed system on the energy level $\mathcal{H}^{-1}(E_V^*)$ keeping fixed the ratio of the frequencies.

• Due to the link between a, e and L, G, this result guarantees that a, e remain close to the unperturbed values within an interval of size of order ε .

Corollary: The values of the perturbed integrals L(t) and G(t) stay close forever to their initial values L_V and G_V and the actual motion (in the mathematical model) is nearly elliptical with osculating orbital values close to the observed ones.